RESEARCH AND

ENGINEERING

OFFICE OF THE UNDER SECRETARY OF DEFENSE



WASHINGTON, D.C. 29301

2 T MAR 1984

MEMORANDUM THRU ASSISTANT DEPUTY UNDER SECRETARY
OF DEFENSE (TACTICAL WARFARE PROGRAMS)

POR CHIEF, SECURITY POLICY AND REVIEW DIVISION

REFERENCE: HASC R&D Subcommittee 1 Mar 84 (afternoon	n)
Attached herewith is(are):	
Corrections (editorial and technical)	
Security Classification	
Answers to Question(s) p.9 line 207	
Insert(s)	
to the Congressional Testimony identified in above cireference. This response is subject to the following	ted caveat(s):
Page determined to be Unclassified Reviewed Chief, RDD, WHS IAW E0 13526, Section 3.5 Date: DEC 0 7 2017 Page determined to be Unclassified WALTER H. SQUIRE Special Assistant DUSD(TWP)	

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Characteristics and Performance of C-58 and C-17A Aircraft

The C-17A is planned for long-term modernization of the The C-1/A is planned for long-term modernization of the Military Afriift Command fleet to augment the airlift capabilities of the C-5A/B, C-141B, and C-130. The C-17 is designed for intertheater and intratheater airlift of military cargo, including outsize and oversize, over intertheater ranges directly to forward area airfields and thereby significantly improves airlift responsiveness. In addition, the C-17A has reduced life cycle costs, reduced manpower requirements, and improved survivability as compared to other comparable aircraft in the fleet.

The C-5R program to produce 50 aircraft was intitieted in early

The C-5B program to produce 50 aircraft was initiated in early FY83 to provide an urgently needed near-term increase in intertheater airlift capability to supplement the C-5A and C-141B

fleet of the Military Airlift Command.

The C-5A was designed in accordance with the Air Force Mission Statement as a basic long range aircraft for rapid airlift of military combat equipment, including outsize and oversize, for direct delivery from COMUS bases or rear marshalling areas to austere airfields in objective areas employing airdrop or airland delivery.

Characteristics and performance of the C-5B and C-17A are summarized in the following table. The current estimates of C-5B operational performance are based on C-5A 1C-5A-1-1 flight handbook data and operations of the rewinged C-5A. The C-17A operational performance estimates are based on calculations, wind tunnel tests of scale models of the C-17A, and contractor tests of the Pratt and Whitney 2037 engine.

The major advantages of the C-17 are:

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1. The ability to use a much higher percentage of the airfields in a selected region of interest than the C-5 can use; this makes the C-17 far more useful for direct delivery.

2. The ability of a force of C-17s to deliver about four times as much cargo per day as a force of C-5s to a forward area airfield typical of those usable by C-17 and C-5 under the ground rules stated. It should be noted that about four times as many C-17s as C-5s would be used to provide the maximum throughput to this airfield since the airfield parking area limits the number of C-5s that can be med.

The major advantage of the C-5 is its ability to deliver larger amounts of cargo of all types per sortie to suitable fields.

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Characteristics and Performance of C-68 and C-17A Aircraft

	C-5B	C-17A
Maximum gross takeoff weight (1b)	769,000	570,000
Maximum payload at 2.25g (1b)	261,000	172,200
Representative deployment payload (1b)1/ Minimum runway width for 1800 continuous	137,800	96,600
turn (ft)	143	134
Minimum runway width for 1800 backup	01	
turn (ft)	m2/	82
Ramp parking area per aircraft without	100 000	100 000
backup (sq. ft)	190,000	100,000
Ramp parking area per aircraft with backup (sq. ft.)	ma2/	60,000
Dackup (Sq. 10.)	3.2	
Cargo offload time (hours)	J.Z	Z
Direct delivery mission example3/:		
Maximum payload delivered to forward		
area airfield at an unrefueled range		
of 2940 ms (1b)	170,000	130,000
Takeoff distance (CONUS airfield)(ft)	7,900	6.700
Landing distance (forward area air-		0,000
field) (ft)	2,000	1,550
Suitable number of airfields (5)4/	23	53
Surface number of diritaries (1)	20	33
Maximum throughput of forward area	***	9. 700
airfield (tons/day)5/	600	2,700

1/ Representative deployment payloads as listed in the Airlift
Master Plan approved by SECAF 29 Sept 1983.
2/ Not applicable; backup capability with reverse thrust not yet
tested or estimated for C-5.
3/ The "direct delivery mission" is one for which both the C-5 and
the C-17 have been designed. The example shown illustrates
relative capabilities of the two aircraft. The ground rules for
this example are:

o All airfield operations (takeoff, landing, taxi,
loading/unloading, and parking) for both C-58 and C-17A are
restricted to paved areas only.

o Takeoff at maximum wartime gross weight from a CONUS major
airfield with payload chosen so that sufficient fuel is
available within this gross weight limit for direct delivery
of the payload 2940 nm from last refueling point to a
forward area airfield, with sufficient fuel remaining for a
500 nm return with zero payload to a rear area airfield.

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o Ranges shown are unrefueled, standard day, zero wind, C-X

o Takeoff distances are ground run length, sea level standard day, zero wind, zero runway slope; C-5B uses 40% flaps, engine air bleed on.

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o Landing distances are ground roll length, sea level standard day, zero wind, zero runway slope; maximum reverse thrust for C-17; no reverse thrust and 100% flaps used for C-5.

o Fuel reserves are based on C-X RFP.

In a selected region of interest, 53% of the airfields examined were suitable for sustained use by C-17 but only 23% were suitable for C-5s because of lack of a taximay or alternatively, inadequate width paved runway for 180° turn around, and/or inadequate paved parking area. (Assumes neither C-17 nor C-5 taxis or parks on unpaved areas of airfields and that C-17 backs up to turn around on narrow runways and to require less space to park at least one aircraft, and C-5 does not back up.)

Based on 50% utilization of an airfield having 468,000 square feet paved parking area (assumes neither C-17 nor C-5 parks on unpaved areas of airfield and that C-17 uses back up and C-5 does not).

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Characteristics and Performance of C-5B and C-17A Aircraft

	C-58	C-17A	
Maximum gross takeoff weight (1b)	769,000	570,000	
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Minimum runway width for 180° backup	240	101	
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Ramp parking area per aircraft without			
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backup (sq. ft.)	NA2/ 3.2	60,000	
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Direct delivery mission example3/:		_	
Maximum payload delivered to forward			
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o Takeoff at maximum wartime gross weight from a CONUS major airfield with payload chosen so that sufficient fuel is available within this gross weight limit for direct delivery of the payload 2940 nm from last refueling point to a forward area airfield, with sufficient fuel remaining for a 500 nm return with zero payload to a rear area airfield.
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o Takeoff distances are ground run length, sea level standard day, zero wind, zero runway slope; C-5B uses 40% flaps, engine air bleed on.

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STRATEGIC DEFENSE INITIATIVE

(The information follows:)

Provided herewith is a copy of the interim charter of the Strategic Defense Initiative Organization.

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STRATEGIC DEFENSE INITIATIVE

DEODET יכוועבוט

Question: Is the Air Force interested in conducting an integrated on-orbit fessibility demonstration of the DARPA Tried Space Laser? If the program proceeds on schedule, when could such test be conducted?

Answer: The DoD continues to pursue technology development activities and studies and analyses that are designed to support a decision in the late 1980's on whether to proceed to a system level demonstration to validate the technology being developed for space based lasers. These activities are described in the 1982 space based lasers. These activities are described in the 1982
Space Laser Program Plan prepared at the request of Congress and approved by the Secretary of Defense in June, 1982. Some changes in the planned activities have been made necessary by the level of funding appropriated in FT 1983 and 1984, but the major demonstrations of technology readiness embodied in the DARPA Triad activities (Alpha — the beam generator, LODE — the beam control subsystem, and Talon Gold — the acquisition, tracking and pointing subsystem) continue on approximately the same schedule as described in the 1982 plan.

These space laser activities are being subsumed into the Strategic Defense Initiative (SDI). In doing so, the DoD will place greater emphasis on the ballistic missile defense mission

Strategic Defense Initiative (SDI). In doing so, the DoD will place greater emphasis on the ballistic missile defense mission application, and will expand the technology efforts schemat to meet the more stringent BHD performance needs and to enhance the relevance of the technology base activities conducted under the Space Laser Program Plan to other SDI activities. Such changes are not expected to have a major impact on the schedule of the baseline technology demonstrations of the Triad, since the Alpha LODE, and Talon Gold are demonstrations of generic technologies

typically identified with

the Triad.

the Tried.
Successful completion of the baseline Tries activities in 1988 could support a decision to proceed with an on-orbit test. In about five years from the decision to proceed (1993), we might expect to have completed the design, fabrication, and flight test of a technology validator based upon the proven designs of the Tried technology demonstrations.

Depending on the results of the current Triad test program, other configurations could

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Quantion: Would as on-orbit test of TALON GOLD of the DARPA TRIAD violate the ANN Treaty? When is a test of TALON GOLD scheduled?

Answer: No. The TALON COLD experiment is part of the BARPA TRIAD program which also includes the ALPHA chemical laser experiment and the Large Optics Demonstration Experiment (LODE). TALON COLD is an experiment, begun in 1979, to be carried on shuttle to explore and validate technologies for very high precision pointing and tracking.

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TAION GOLD has been included under the Strategic Defense Initiative. However, TAION GOLD was formulated, and continues to be aimed at generic pointing and tracking technology. The TAION GOLD program is predicated on a number of national requirements including precision pointing and tracking used for national technical means of verification and NASA scientific experiments. TAION GOLD is not a component in any space-based weapon system, ABN or otherwise.

Such experiments are fully compliant with all US treaty commitments including the ABM Treaty and the Outer Space Treaty. The initial flight test of TALON GOLD is scheduled for 1988.

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Mine Hunting Technology



Congressman Bennett: On another subject entirely, I wonder if there are any breakthroughs, or any efforts being made to get a better way of eliminating mines from the sea?

In other words, laser or other things being utilized are looked at. It strikes we that we don't operate in this field in a very advanced manner. Are they being done in that field? Does anybody know about it?

Answer: There have been no recent technological breakthroughs in the field of mine detection/disposal. Tests have been made to evaluate lasers for various missions and transmissivity in clear sea water makes them reasonable candidates for some functions. Unfortunately the coastal waters where mines are often encountered are characterized by extreme turbidity, especially at or near the bottom. This condition severely limits the utility of laser systems or underwater television systems to tens of feet. Our program for the detection and neutralization of mines is robust and balanced. It includes the development of a new generation acoustic mine hunting system for the recently authorized classes of mine hunting ships, as well as procurement of the recently developed mine neutralization vehicle (MNV). The MNV includes both acoustic and underwater television capability. In addition, the Mavy is pursuing a program of channel mapping and clearance to eliminate or map mine-like objects in approaches to selected US ports. While we remain alert to the possibility of a technological breakthrough, we are following a strategy of a balanced evolutionary approach to this critical problem.

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Action officer/extension Mr. Robert P. Belaney, x65531			19 March 1984
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